

International Center for
Quantum-field Measurement Systems for
Studies of the Universe and Particles

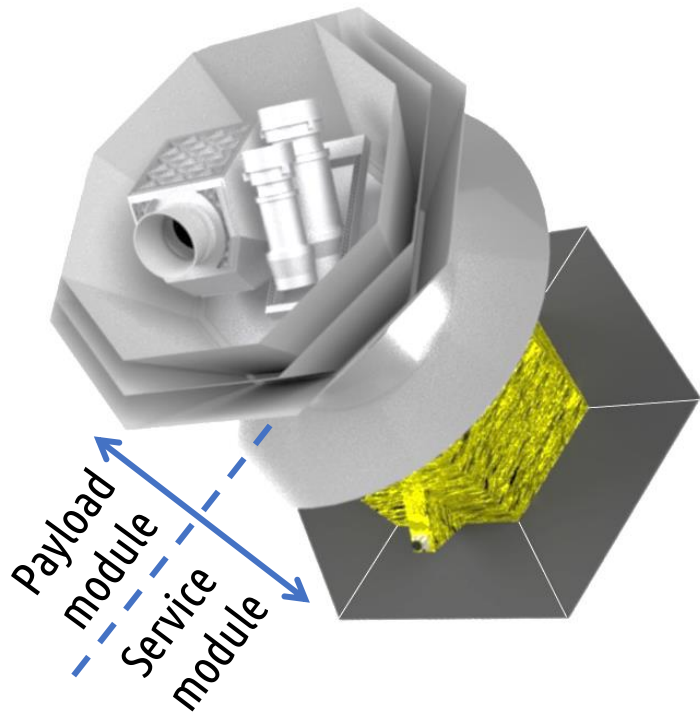
Multi-disciplinary design of Focal Plane Sub-System of LiteBIRD Low Frequency Telescope

Daisuke Kaneko, Systemology Support Section,
on behalf of QUP SSS and CMB group

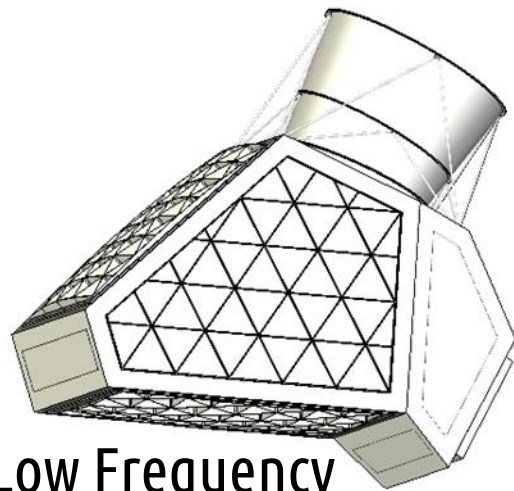




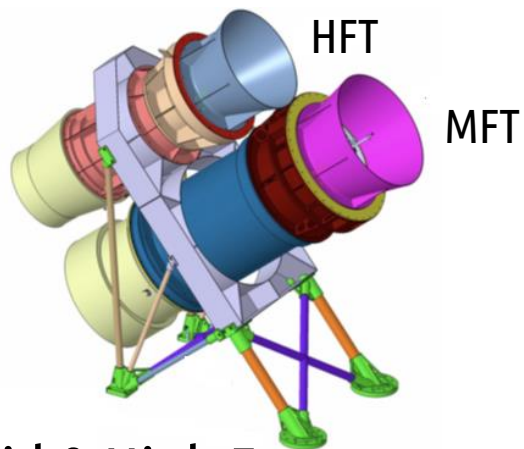
QUP apportionment of LiteBIRD system



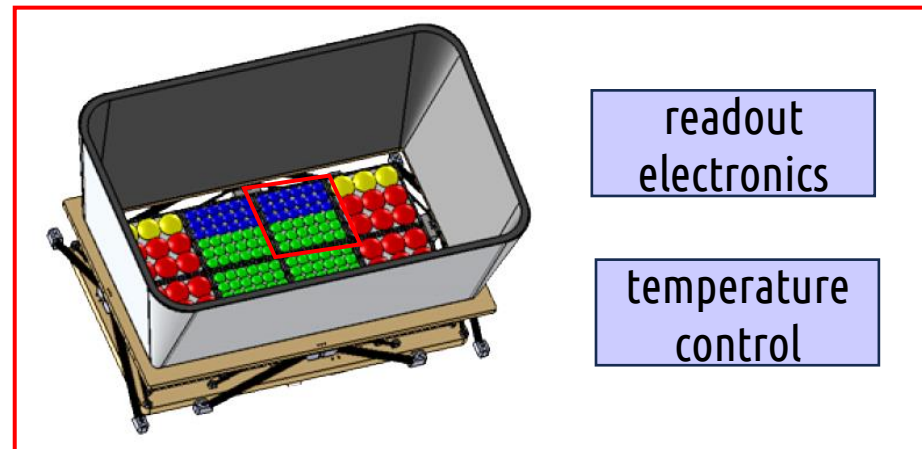
LiteBIRD satellite



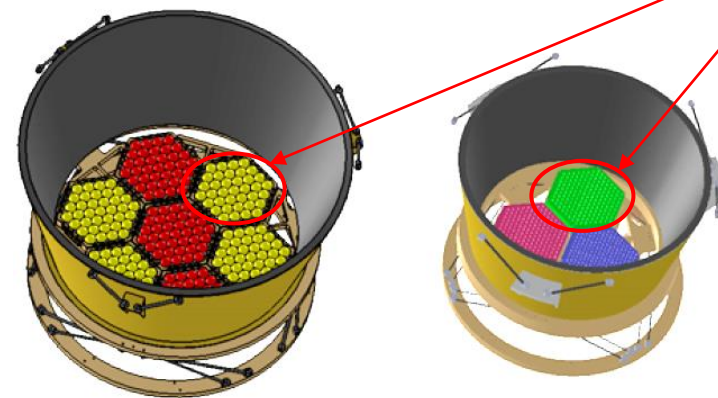
Low Frequency Telescope (LFT)



Mid & High Frequency Telescope (MHFT)



Focal Plane Sub-system (FPSS)

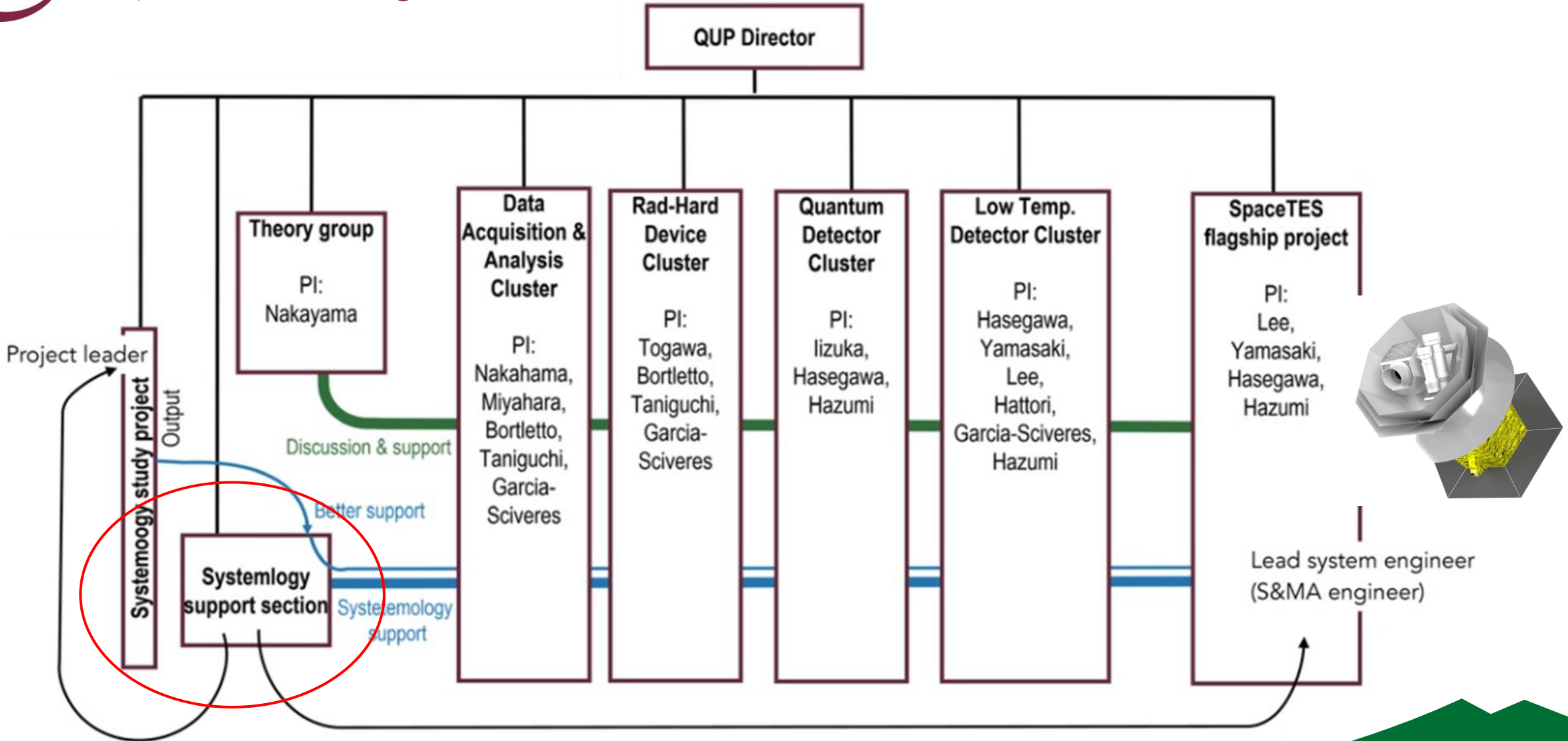


Focal Plane Unit of MFT & HFT






QUP and Systemology Support Section (SSS)





SSS organization

Section leader
Mitsuda



Systems Engineering unit leader
Mitsuda

System Engineer
Kaneko



System Engineer
Shiomi



Principal Engineer
Kagawa



S&MA
from HIREC company

System Engineer
Mori (from Toyota)

Concurrently assigned as the lead system engineer and the S&MA engineer of SpaceTES (LiteBIRD QUP) project, respectively.

Concurrently assigned as the project leader of the systemology study project

S&MA = System safety & Mission Assurance

(C) Senior members who are responsible to foster young system architects.

Multi-discipline Engineering unit leader
Narasaki

3D CAD designer
Hirasawa

Mechanical/thermal Engineer
Part-time assignment from KEK

Electronics engineer
Part-time assignment from KEK



Systems Engineering of FP Sub-system

In design of FPU,
the input is requirements, not only directly for
observation, but the all items must also be integrated.

SSS aids in finding realistic design solutions through
engineering, involving experts from various disciplines
such as

- Mechanical
- Electrical
- Cryogenic
- etc.

These disciplines often interrelate.

Types of requirements to consider

Functional

Performance

Interface

Operational

Environmental

Reliability

Safety

Other



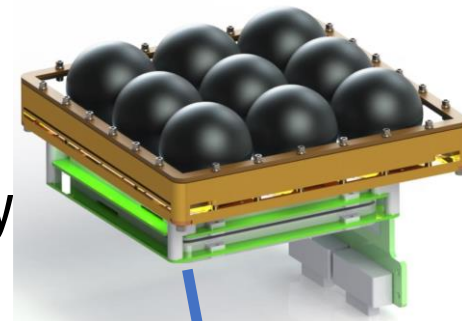
Topics in this talk

- (1) Structural design and vibrational analysis
- (2) Faraday cage to protect from EMI
- (3) Our current FPU design



Basic design of LFT FPU

Focal plane subassembly (detector)



Detector hood (at 1.8K)

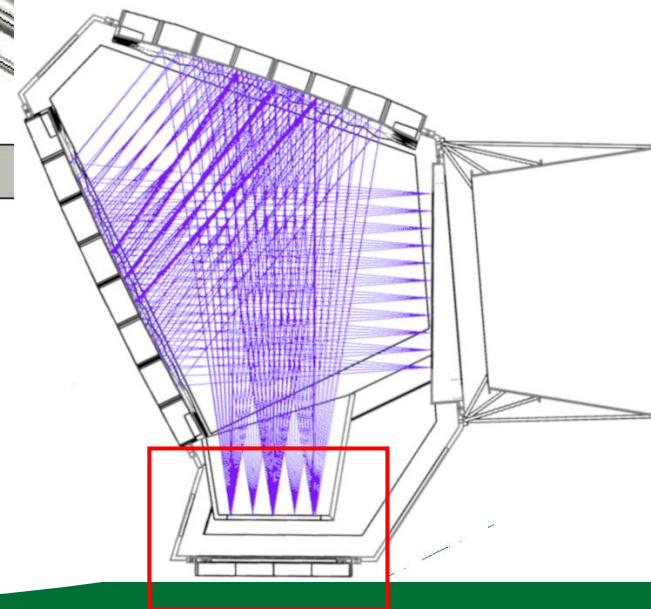
1.8K stage

100mK stage

CFRP struts between different temperature stages

4.8K mechanical I/F point to LFT structure

300mK stage





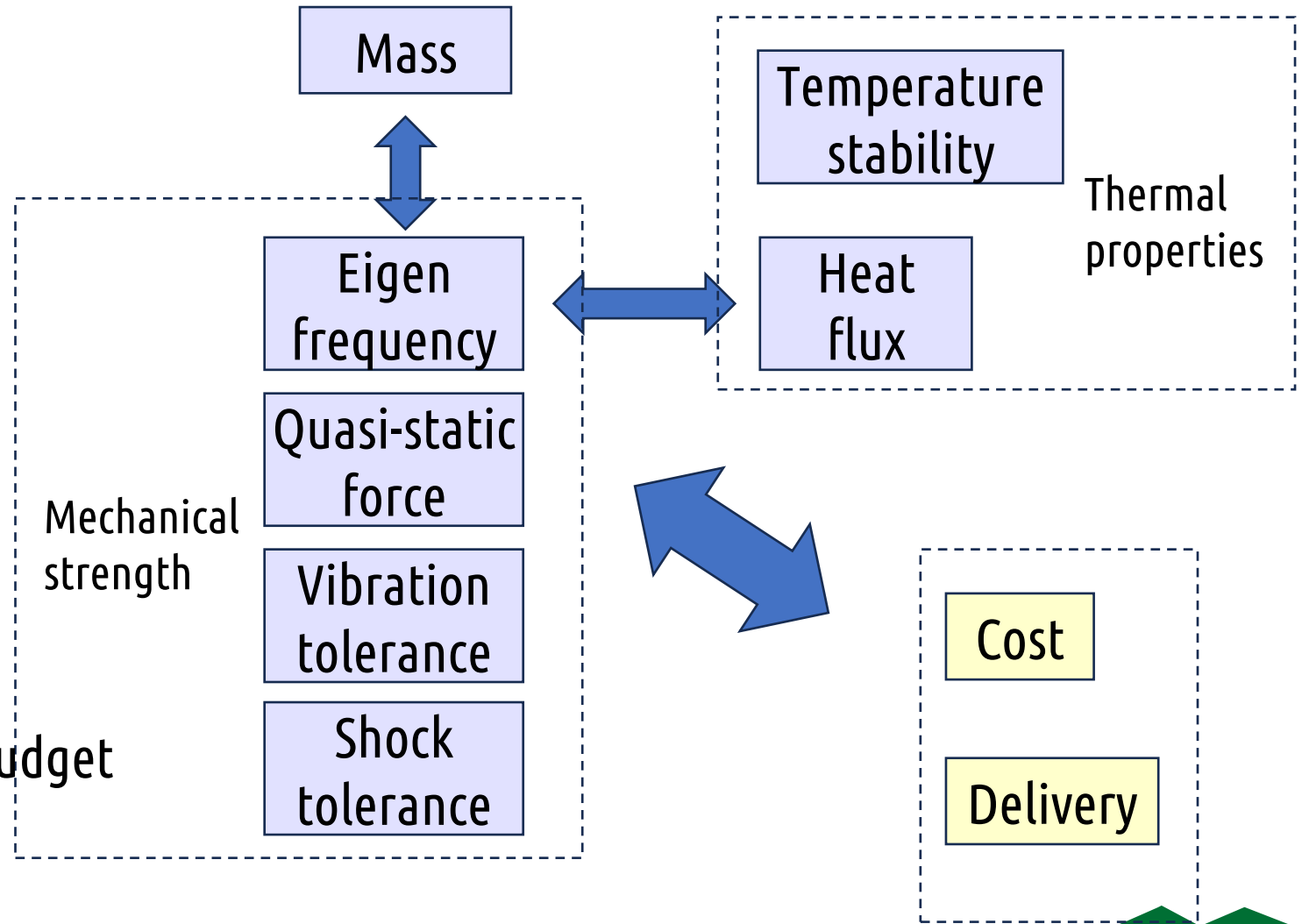
Requirements and conditions

There are lots of trade-offs in the design of the system.

Ex. "Minimum eigen frequency of FPU > 141Hz"

To meet this requirement,
shorter or thicker CFRP struts
increase heat flux
stronger stage structure
increase mass
cost and delivery to keep within budget

If no feasible solution can be found,
need redesign (of subsystem or parent system)





Structural analysis

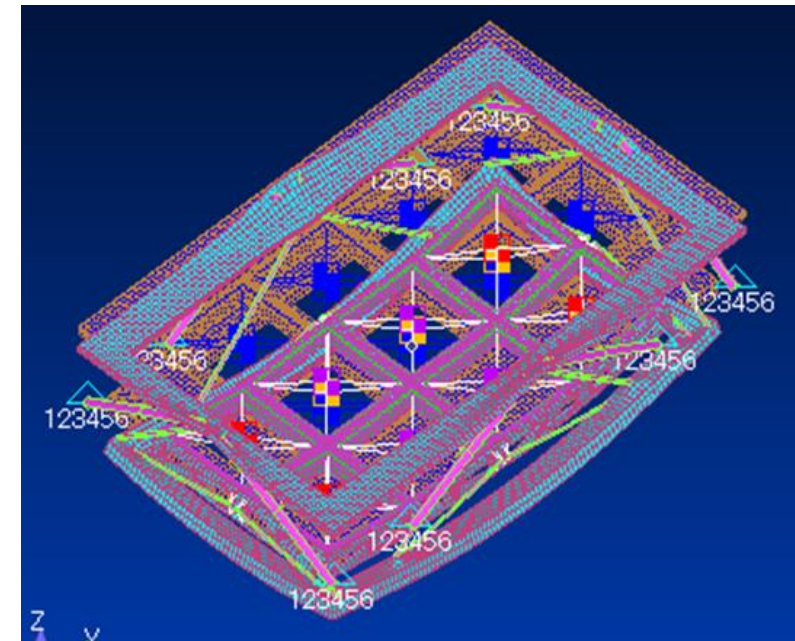
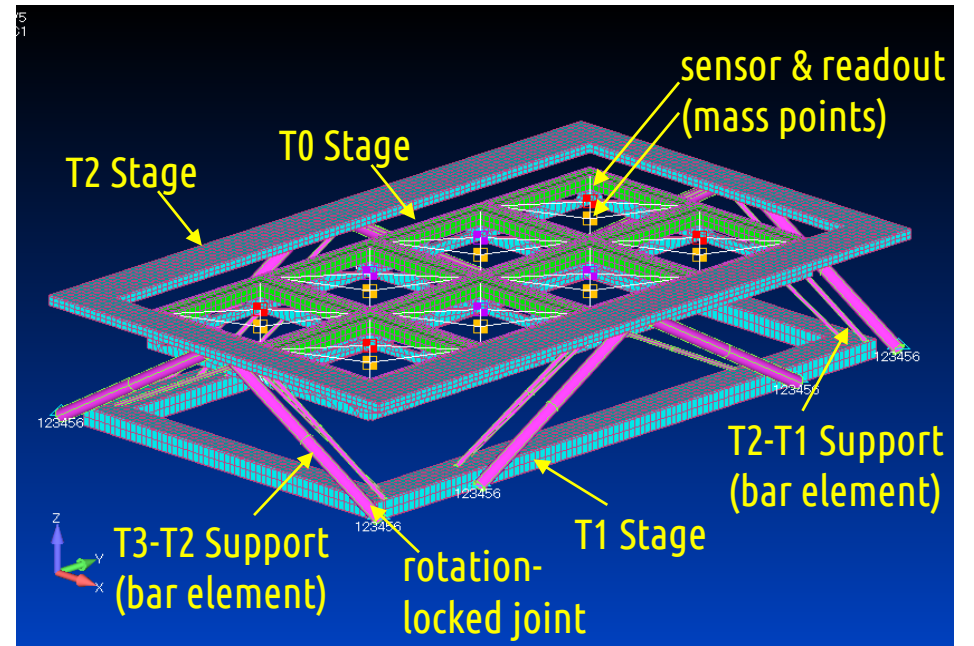
We prepare to model and analyze structural analysis with NASTRAN.

A preliminary calculation was performed on the initial design of the FPU.

We demonstrated the reliable results by comparing with the results from the manufacturing company.

We found that this version and some simple modification didn't meet frequency requirements.

We will start analysis with our updated FPU





Optimization of structures

We plan to optimize design on simulation

- (1) Dimension of stage, position, length of CFRP etc.
 - (2) Topological optimization of structure also in scope
- Asking support of external specialist

Performance of CFRP struts is one of the critical point

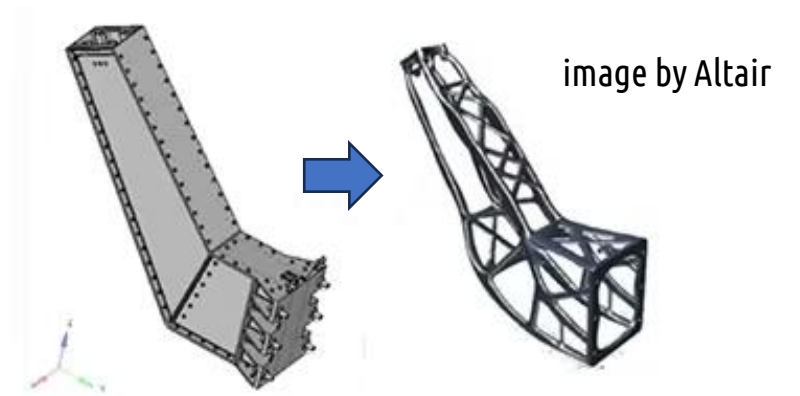
Thermal conductance and its dependence

on the choice of material is known precisely enough

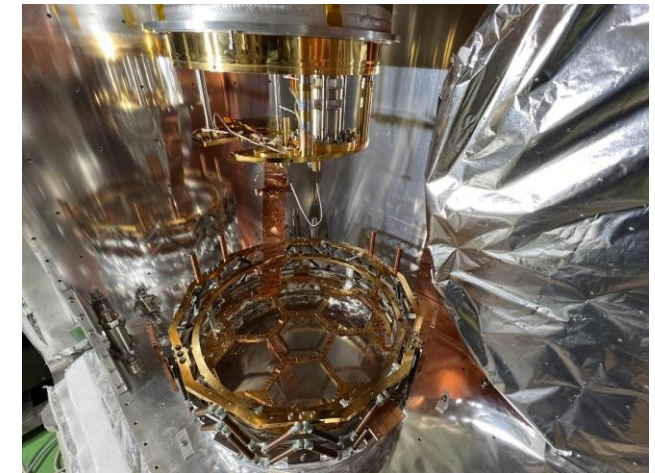
Prototype measurement at QUP laboratory is planned

Reinforcement of CFRP against launch stress is

planned too



An image of topological optimization



Cooling test image with dilution refrigerator cryostat at QUP lab.



Electromagnetic integrity

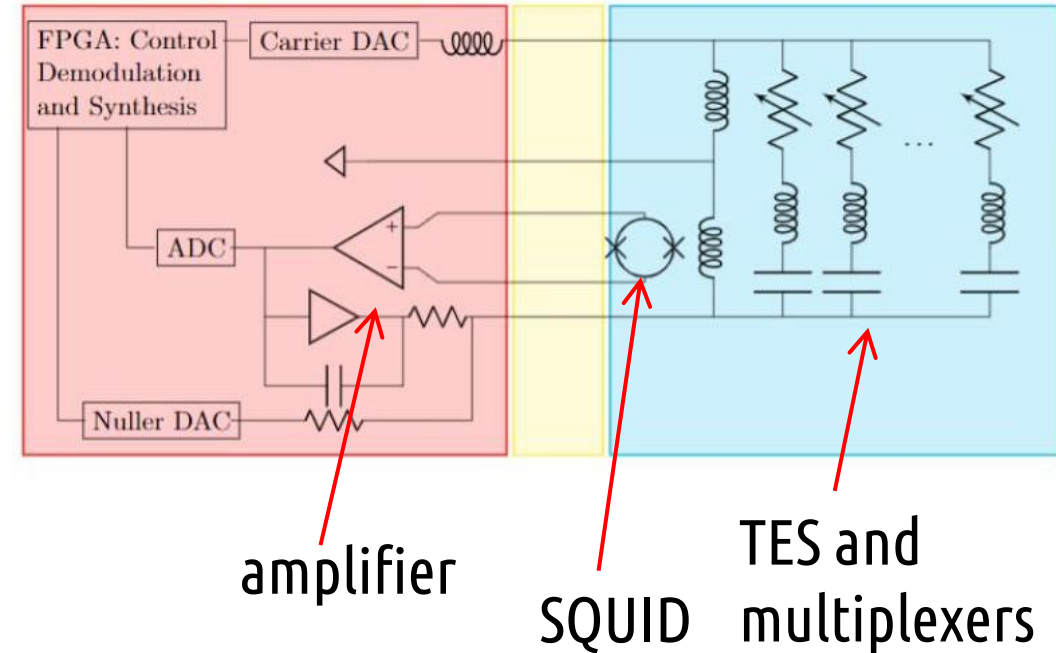
Electromagnetic requirements:

- Tolerance to DC and AC magnetic field
- Compliance with EMI standards
- Grounding scheme must follow required structure
- etc.

In addition, as we adopt SQUID readout scheme,

- SQUID must be inside of Faraday cage to protect from outside RF noise
 - ← there are readout bundle and heat link inside of the vacuum shell
 - ↑ mechanism to close cage
 - ↑ readout bundle and heat link to pass through
 - ← additional thermal pass appears
 - ↑ thermal budget must be reconsidered

Readout schematic image





Fundamental idea of LFT Faraday cage

Sky side

Design after ground CMB experiments PB/PB-2 receiver

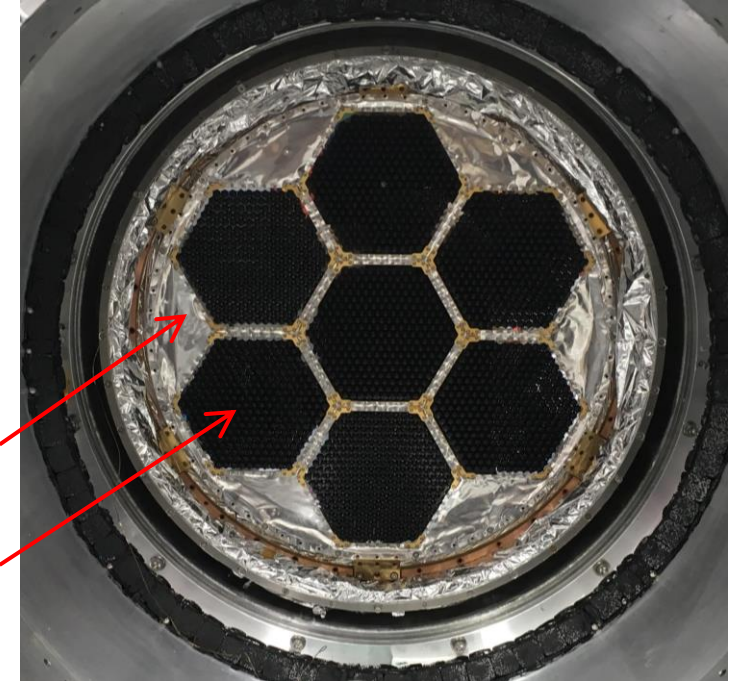
Material

Aluminized Mylar film

Nb deposited on wafer

Anti-sky side

Cannot be like PB/PB-2, as readout shares vacuum chamber
add shell structure of Faraday cage
at T2 to surround SQUID circuits



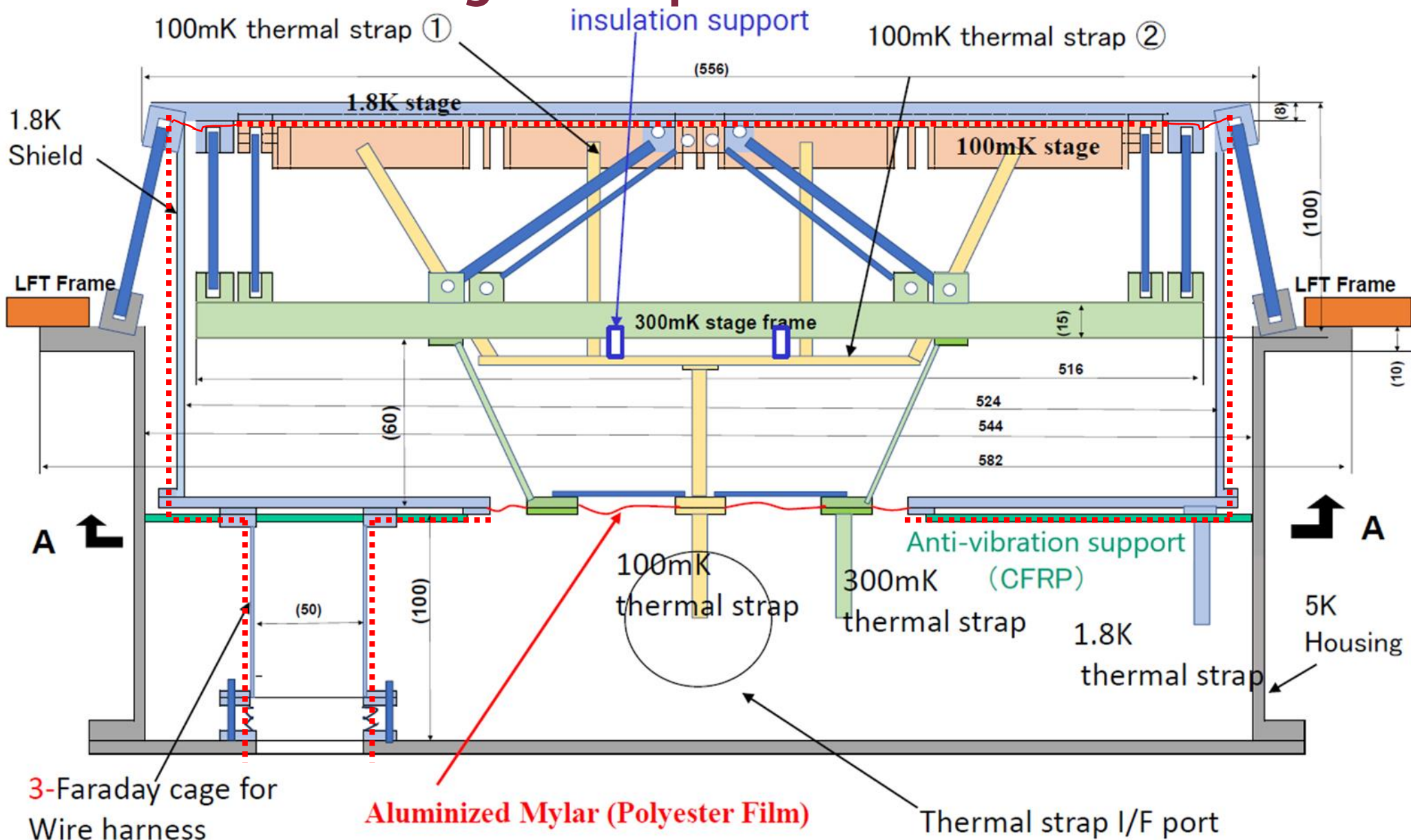
Aluminized Mylar

Wafer face is part
of shield

PB-2 focal plane

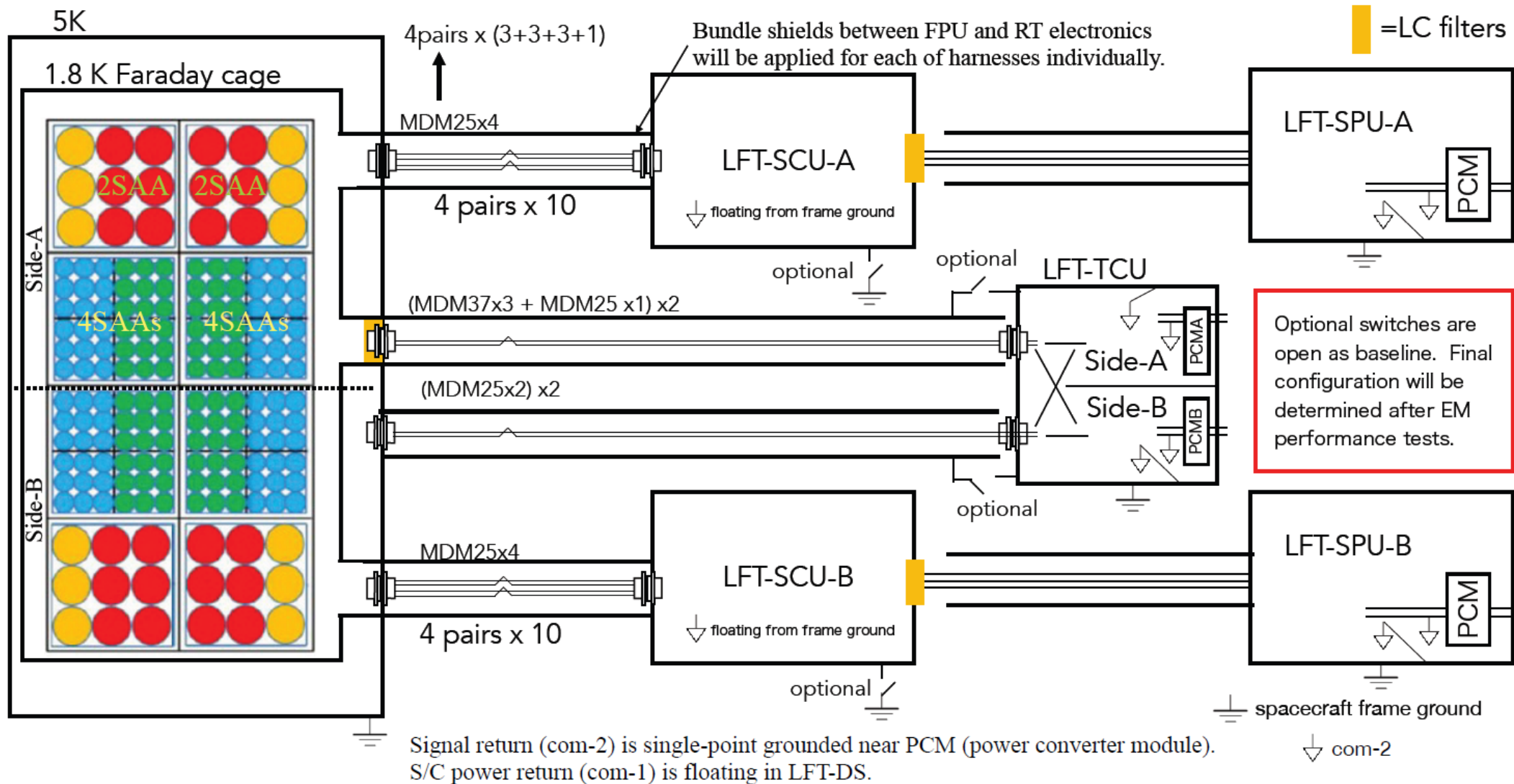


Schematic image of updated FPU



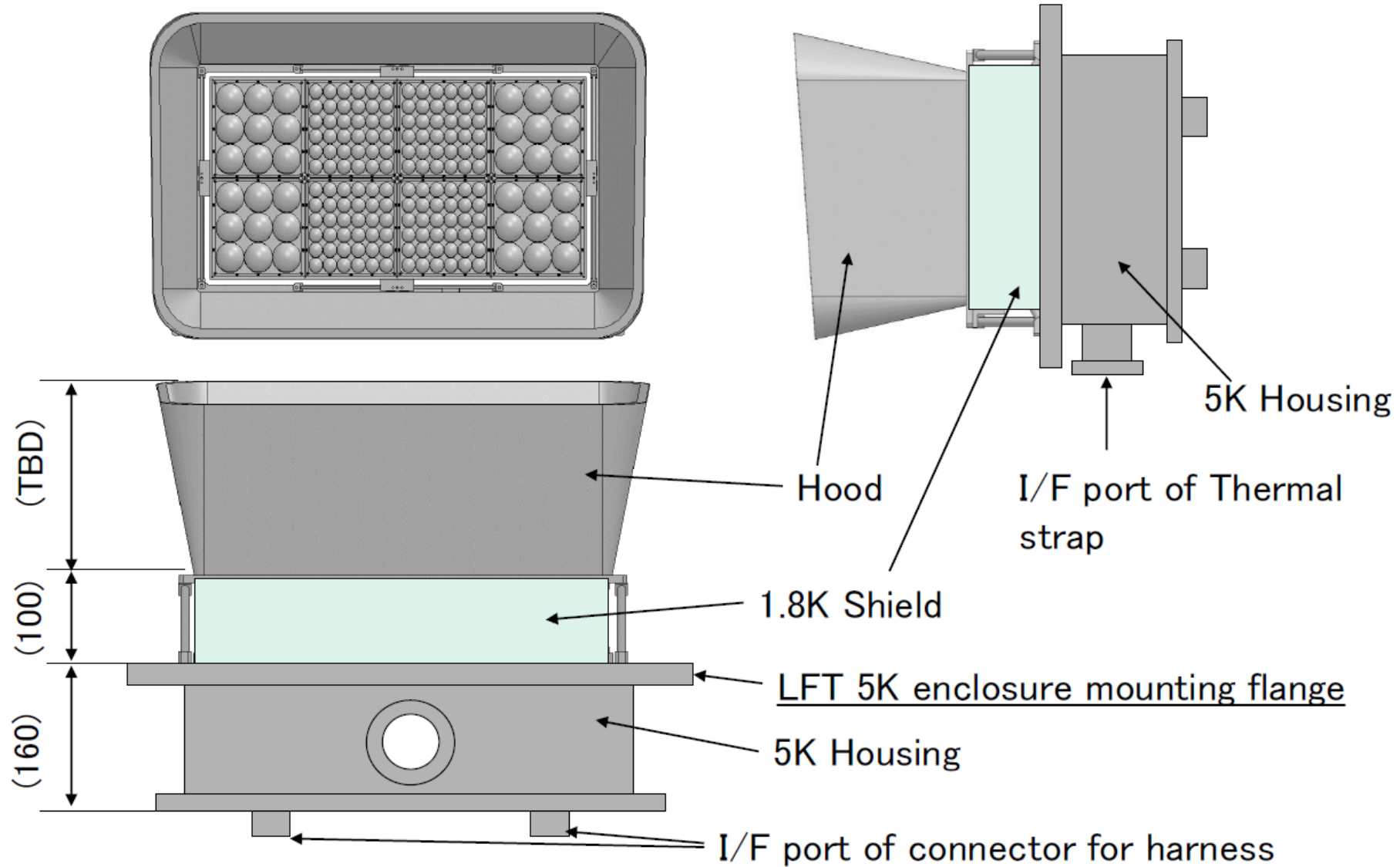


Grounding scheme plan



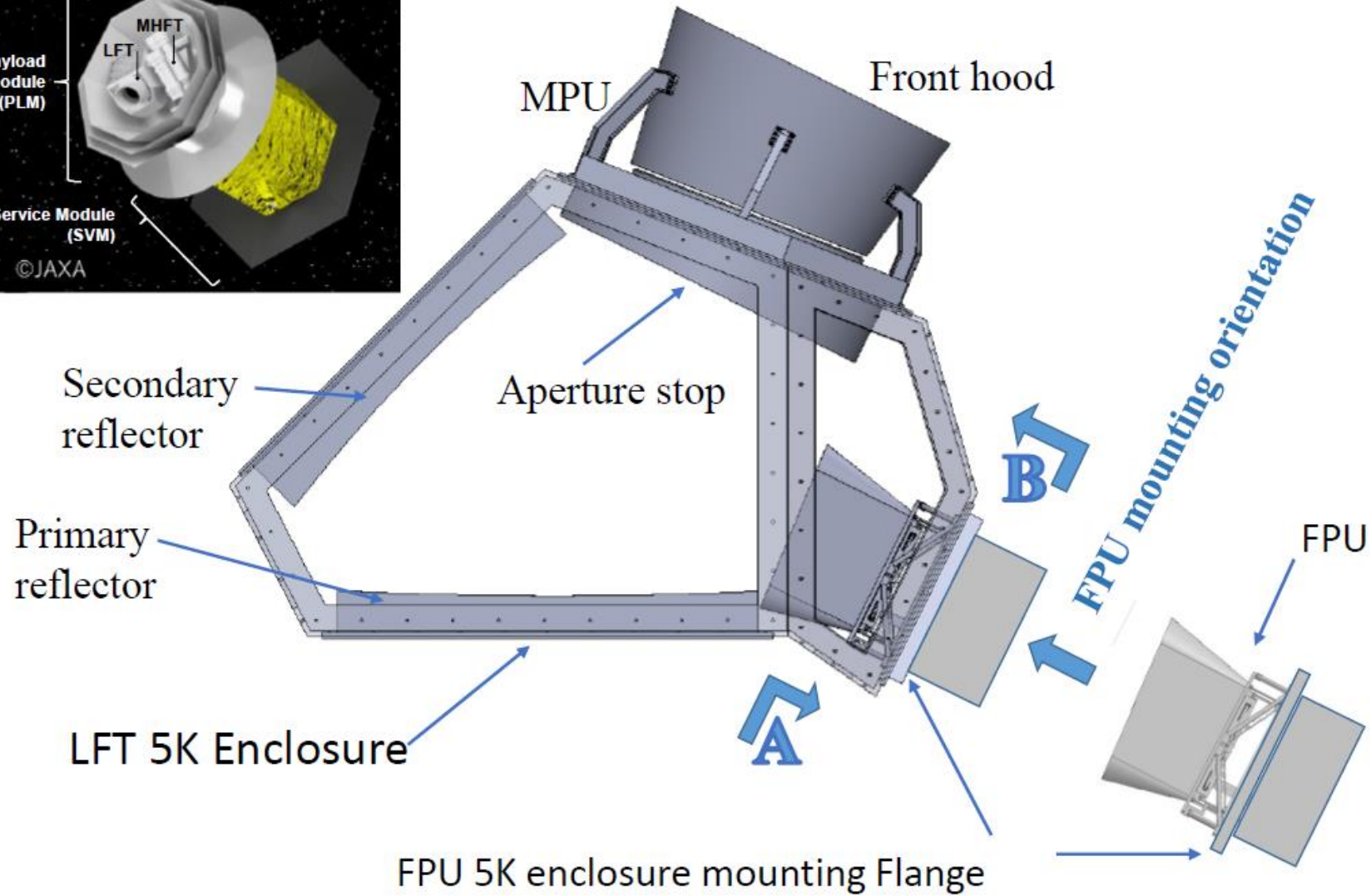
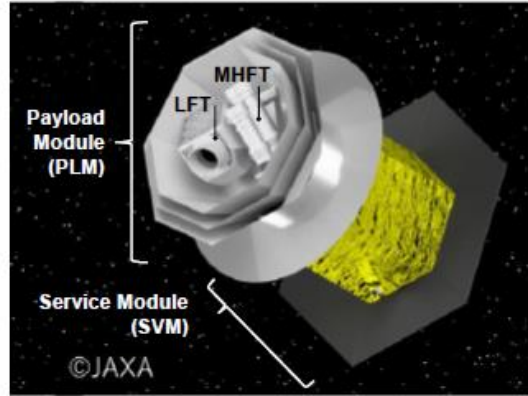


Current plan of FPU structure



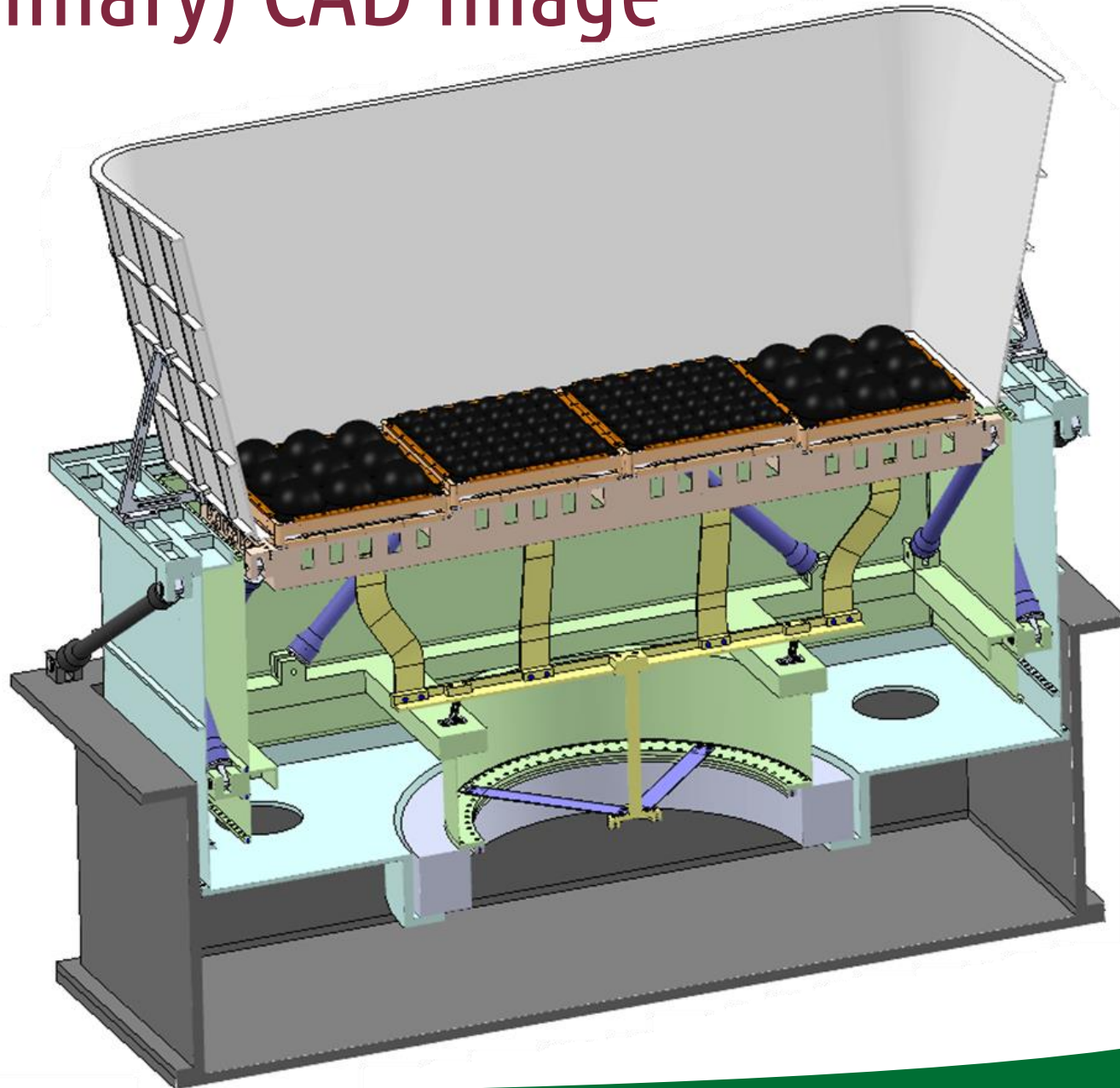


Mounting plan onto LFT





(preliminary) CAD image





Summary

QUP is in charge of design and production of FPSS.

Many challenges in design of FPU.

Systemology support section is establishing a multi-disciplinary design infrastructure in QUP.

FPU design is now in progress with supports of experts of different disciplines.

Design integrity will be confirmed with simulations and prototyping tests.